

Equation of Trajectory

From radial component we get $\ddot{r} - r\dot{\theta}^2 + \frac{\mu}{r^2} = 0$

$$\dot{r} = \frac{dr}{d\theta} \frac{d\theta}{dt} = \dot{\theta} \frac{dr}{d\theta} = \frac{h}{r^2} \frac{dr}{d\theta} = -h \frac{d}{d\theta} \left(\frac{1}{r} \right) \quad \ddot{r} = \frac{d\dot{r}}{dt} = \frac{d\dot{r}}{d\theta} \frac{d\theta}{dt} = \frac{h}{r^2} \frac{d\dot{r}}{d\theta} = -\frac{h^2}{r^2} \frac{d^2}{d\theta^2} \left(\frac{1}{r} \right)$$

$$-\frac{h^2}{r^2} \frac{d^2}{d\theta^2} \left(\frac{1}{r} \right) - \frac{h^2}{r^3} + \frac{\mu}{r^2} = 0$$

$$\frac{d^2}{d\theta^2} \left(\frac{1}{r} \right) + \frac{1}{r} - \frac{\mu}{h^2} = 0 \quad \text{let } \frac{1}{r} - \frac{\mu}{h^2} = Z \quad \Rightarrow \frac{d^2 Z}{d\theta^2} + Z = 0$$

$$Z = A \cos \theta + B \sin \theta = C \cos(\theta - \psi), \quad \text{where } C = \sqrt{A^2 + B^2} \text{ and } \psi = \tan^{-1} \left(\frac{B}{A} \right)$$

$$\frac{1}{r} = \frac{\mu}{h^2} + C \cos(\theta - \psi) = \frac{\mu}{h^2} (1 + e \cos(\theta - \psi)) \quad \text{where } e = \frac{Ch^2}{\mu}$$

The equation of trajectory of mass 'm' about 'M' is given by: $r = \frac{h^2/\mu}{(1 + e \cos(\theta - \psi))}$

The equation above represents a class of curves referred to as conic curves. We take a look at the mathematical description of the conic curves next.

The conic section

The family curves obtained when a plane cuts a conical surface is called conics.

The different conics are shown below.

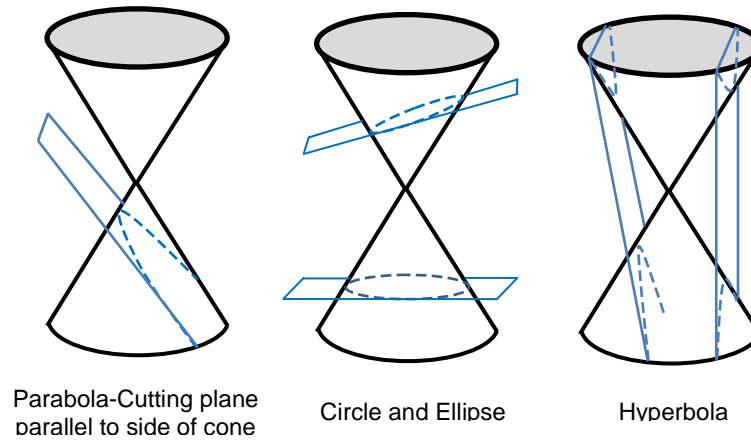


Figure 1

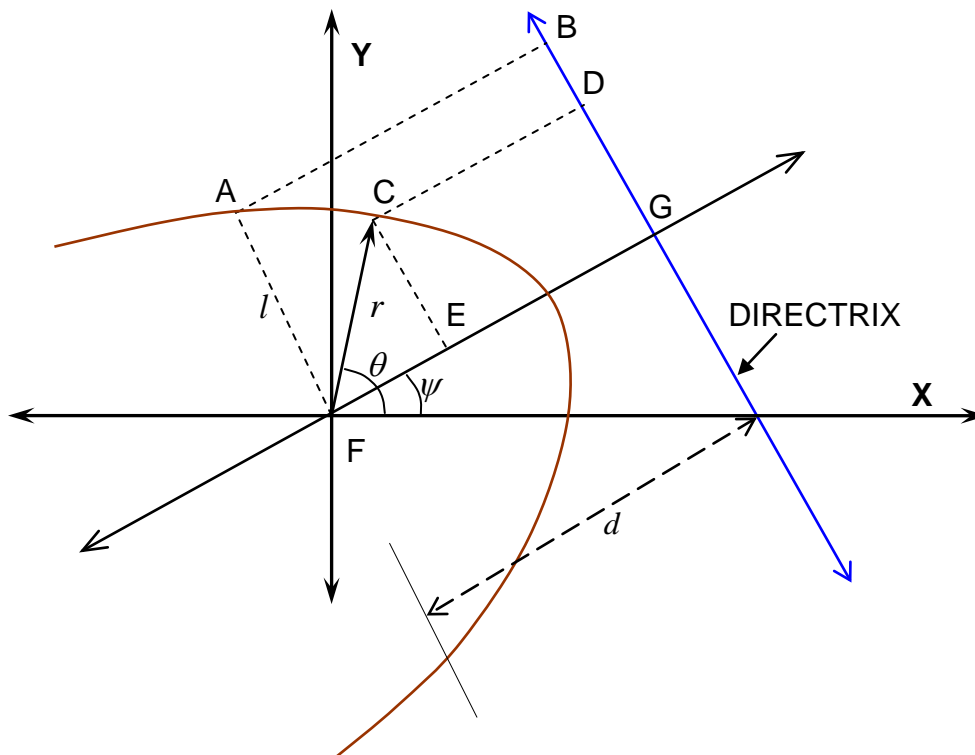


Figure 2

Mathematically defining, a conic is a locus of a point which moves so that the ratio of its distance from a fixed point to its distance from a fixed line is a positive constant. The ratio is the eccentricity of the conic, the fixed point the focus and the fixed line the directrix (Fig. 2).

$$r = CF = e CD = e (FG - FE) = e (AB - FE) = l - e r \cos(\theta - \psi)$$

$$r = \frac{l}{1 + e \cos(\theta - \psi)} = \frac{ed}{1 + e \cos(\theta - \psi)}$$

In the cartesian coordinate system, the graph of a quadratic equation in two variables is always a conic section, and all conic sections arise in this way. If the equation is of the form

$$ax^2 + 2cxy + by^2 + 2dx + 2ey + c = 0$$

then:

if $c^2 = ab$, the equation represents a parabola

if $c^2 < ab$ and $a \neq b$ and/or $c \neq 0$, the equation represents an ellipse

if $c^2 > ab$, the equation represents a hyperbola

if $c^2 < ab$ and $a = b$ and $c = 0$, the equation represents a circle

Trajectory Classification

Equation of trajectory for a two body problem $r = \frac{h^2/\mu}{(1 + e \cos(\theta - \psi))}$

Let $\psi = 0$, then $r = \frac{h^2/\mu}{1 + e \cos \theta}$ (see figure below)

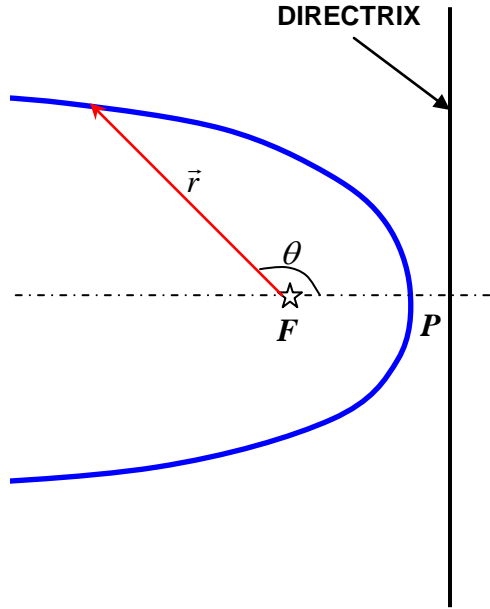


Figure 3

$$\underbrace{\left(\frac{1}{2} \dot{r}^2 + \frac{1}{2} r^2 \dot{\theta}^2 \right)}_{\text{kinetic energy / mass}} + \underbrace{\left(-\frac{\mu}{r} \right)}_{\text{potential energy / mass}} = \text{const.} = \underbrace{E_T}_{\text{total energy / mass}} \quad (\text{Orbital Energy})$$

or $E_T = \frac{1}{2} V^2 - \frac{\mu}{r} = \text{const.}$, where $V^2 = \frac{1}{2} \dot{r}^2 + \frac{1}{2} r^2 \dot{\theta}^2$

Now the constant value of E_T can be evaluated at $\theta = 0$, where $r = r_{\min}$ and

$$\dot{r} = 0$$

$$E_T = \frac{1}{2} r_{\min}^2 \dot{\theta}^2 - \frac{\mu}{r_{\min}}, \text{ also } h = r^2 \dot{\theta} = \text{const.} = r_{\min}^2 \dot{\theta} \text{ and } r_{\min} = \frac{h^2/\mu}{1 + e}$$

$$\therefore E_T = \frac{1}{2} r_{\min}^2 \dot{\theta}^2 - \frac{\mu}{r_{\min}} = \frac{1}{2} \frac{h^2}{r_{\min}^2} - \frac{\mu}{r_{\min}} = \frac{\mu(e-1)}{2r_{\min}} = -\frac{\mu^2(1-e^2)}{2h^2}$$

$$E_T = -\frac{\mu^2(1-e^2)}{2h^2} = \frac{1}{2} r^2 \dot{\theta}^2 - \frac{\mu}{r}$$

When,

$$e < 1 \text{ then } E_T < 0 \quad -\frac{\mu}{r} \left(\frac{\text{Potential Energy}}{\text{mass}} \right) > \frac{1}{2} V^2 \left(\frac{\text{Kinetic Energy}}{\text{mass}} \right), \text{ the}$$

trajectory is an ELLIPSE (Kepler's first law)

$$e = 1 \text{ then } E_T = 0 \quad -\frac{\mu}{r} \left(\frac{\text{Potential Energy}}{\text{mass}} \right) = \frac{1}{2} V^2 \left(\frac{\text{Kinetic Energy}}{\text{mass}} \right), \text{ the}$$

trajectory is a PARABOLA

$$e > 1 \text{ then } E_T > 0 \quad -\frac{\mu}{r} \left(\frac{\text{Potential Energy}}{\text{mass}} \right) < \frac{1}{2} V^2 \left(\frac{\text{Kinetic Energy}}{\text{mass}} \right), \text{ the}$$

trajectory is a HYPERBOLA

Further Reading:

1. Mechanics & Thermodynamics of Propulsion (2nd Ed.) by Philip G. Hill and Carl Peterson (2nd Edition) Chapter #10 (section 10.5)
2. Introduction to Space Flight by Francis J. Hale (Chapter #3) Prentice Hall.
3. Space Flight Dynamics William E Wiesel (2nd Edition) Chapter #3.